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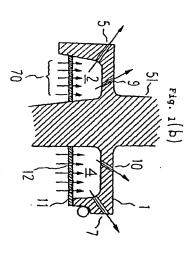
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Gas turbine rotor blade platform

1 inclinedly upwardly. Cooling air 70 flows into the cav-4 and all these cooling holes pass through the platform ing hole 6 with cavity 3 and cooling holes 7, 8 with cavity 2, 3, 4. Cooling hole 5 communicates with cavity 2, coolimpingement plate 11 being provided below the cavities provided. Cavities 2, 3, 4 are formed in platform 1 with ple cooling structure and effecting uniform cooling is Gas turbine moving blade platform having sim-

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cooling passage is provided and workability is enform 1 is cooled uniformly, no lengthy and complicated cooling platform 1 peripheral portions. Thus, the plating holes 5, 6, 7 to blow outside inclinedly upwardly for portion. The cooling air 70 further flows through the cool-11 for effecting impingement cooling of platform 1 plane ities 2, 3, 4 through holes 12 of the impingement plate



of For more details about this annex : see Official Journal of the European Patent Office, No. 12/82

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European Patent Office

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Application Number EP 99 10 3381

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LACK OF UNITY OF INVENTION SHEET B

The Search Division considers that the present European patient application does not comply with the requirements of unity of inventions and relates to several inventions or groups of inventions, namely:

1. Claims: 1-4

gas turbine moving blade platform comprising a cavity formed in said platform around a base portion of the moving blade for introducing thereinto a cooling air; and a plurality of cooling holes communicating with said cavity and opening at a peripheral end surface of said platform

2. Claim: 5

gas turbine moving blade platform comprising two cooling passages on each side of the moving blade, communicating at the one end with a leading edge passage of the moving blade and having at its the other end an opening at a side end surface of said platform; a cover for closing said opening of each of said two cooling passages; and at least three linearly formed cooling passages in said platform, each communicating at its one end with any one of said two cooling passages and having at its the other end an opening at a rear end surface of said platform

3. Claim : 6

gas turbine moving blade platform consisting of an upper platform and a lower platform, a cavity between said upper platform and lower platform on each side of ventral and dorsal sides of the moving blade, and characterized in comprising a cooling passage, being bored in said upper platform along each of both side portions of said upper platform, communicating at its one end with said cavity at a front portion of said platform and having at its the other end an opening at a rear end surface of said platform; and a multiplicity of cooling holes, being bored in said lower platform and passing through upwardly into said cavity thereabove from a bottom surface of said lower platform